

IGNITION SYSTEM: EXHAUST MANIFOLD:	1800 9.4:1 90 210 TA V+OC, AC CDIS WC XTALYST	WIT	FUEL PRESS. R MIN. METHANE RATED ALTITUE AT AIR TO TURE EXHAUST O2 EN	SYSTEM:LPG IMPTOMER SUPPLIED AIR FUEL RATIO CONTEPRESS. RANGE (PSIG):1.5 -METHANE NUMBER:D ALTITUDE (FT):4R TO TURBO. TEMP. (°F):UST O2 EMISSION LEVEL:0.4 %LHV (BTU/SCF):4		
RATING AND EFFICIENCY	NOTES	LOAD	100%	75%	50%	
ENGINE POWER (WITHOUT FAN)	(1)	BHP	365	274	182	
GENERATOR POWER (WITHOUT FAN)	(2)	EKW	250	187	125	
ENGINE EFFICIENCY (ISO 3046/1)		%	35.4	33.7	31.4	
ENGINE EFFICIENCY (NOMINAL)		%	35.4	33.7	31.4	
THERMAL EFFICIENCY (NOMINAL)		%	50.5	52.3	55.5	
TOTAL EFFICIENCY (NOMINAL)		%	85.9	86.0	86.9	
ENGINE DATA						
FUEL CONSUMPTION (ISO 3046/1)	(6)	BTU/bhp-hr	7182	7551	8093	
FUEL CONSUMPTION (NOMINAL)	(6)	BTU/bhp-hr	7182	7551	8093	
AIR FLOW (77 °F, 14.7 psi)	(7)	SCFM	474	375	269	
AIR FLOW	(7)	lb/hr	2103	1661	1191.00	
COMPRESSOR OUT PRESSURE	. ,	in. HG (abs)	53.2	50.7	41.8	
COMPRESSOR OUT TEMPERATURE		°F	217	203	158	
AFTERCOOLER AIR OUT TEMPERATURE		°F	95	92	91	
INLET MAN. PRESSURE	(8)	in. HG (abs)	52	41.3	30.5	
INLET MAN. TEMPERATURE (MEASURED IN PLENUM)		°F)	97	95	93	
TIMING	(10)	°BTDC	30	30	30	
EXHAUST STACK TEMPERATURE	(11)	°F	1038	998	960	
EXHAUST GAS FLOW (@ stack temp.)	(12)	CFM	1485	1142	797	
EXHAUST MASS FLOW	(12)	lb/hr	2234	1765	1265	
<u> </u>			n			
EMISSIONS DATA						
NOx (as NO2)	(13)	g/bhp-hr	13.56	14.39	14.86	
СО	(14)	g/bhp-hr	13.56	14.39	14.86	
THC (molecular weight of 15.84)	(14)	g/bhp-hr	2.36	2.71	3.12	
NMHC (molecular weight of 15.84)	(14)	g/bhp-hr	0.35	0.41	0.47	
EXHAUST O2	(15)	% DRY	0.4	0.4	0.4	
LAMBDA			1.00	1.00	1.00	
	7					
HEAT BALANCE DATA			1			
	(16)	BTU/min	43662	34428	24601	
HEAT REJECTION TO JACKET (JW)	(17) (22)	BTU/min	12803	10937	8672	
HEAT REJECTION TO ATMOSPHERE	(18)	BTU/min	1746	1377	984	
HEAT REJECTION TO LUBE OIL (OC)	(19) (22)	BTU/min	2024	1729	1371	
HEAT REJECTION TO EXHAUST (LHV to 77°F)	(20)	BTU/min	10441	7939	5480	
HEAT REJECTION TO EXHAUST (LHV to 350°F)	(20)	BTU/min	7200	5344	3600	
HEAT REJECTION TO A/C (AC)	(21) (23)	BTU/min	1179	844	359	

### CONDITIONS AND DEFINITIONS

ENGINE RATING OBTAINED AND PRESENTED IN ACCORDANCE WITH ISO 3046/1STD. REF. CONDITIONS OF 77°F, 29.6 IN HG BAROMETRIC PRESSURE, 500 FT ALTITUDE). NO OVERLOAD PERMITTED AT RATING SHOWN. CONSULT ALTITUDE CHARTS FOR APPLICATIONS ABOVE MAXIMUM RATED ALTITUDE AND/OR TEMPERATURE.

EMISSION LEVELS ARE BASED ON THE ENGINE OPERATING AT STEADY STATE CONDITIONS. EMISSION TOLERANCES SPECIFIED ARE DEPENDANT UPON FUEL QUALITY. METHANE NUMBER CANNOT VARY MORE THAN ± 3. PUBLISHED PART LOAD DATA REQUIRES CUSTOMER SUPPLIED AIR FUEL RATIO CONTROL.

ENGINE RATING IS WITH 2 ENGINE DRIVEN WATER PUMPS.

FOR NOTES INFORMATION CONSULT PAGE THREE.

G3406

## GAS ENGINE TECHNICAL DATA

## **CATERPILLAR®**

		FUE	L USAG	e guid	E									
CAT M	ETHANE	NUMBER	30	35	40	45	50	55	60	65	70	75	80	85-100
IGNITION TIMING		15	16	17	18	19	20	21	22	24	27	29	29	
DERATION FACTOR 0.77 0.77					0.77	0.77	0.77	0.77	0.77	0.77	1.00	1.00	1.00	1.00
					CTOPS			1						
	A	LTITUDE	DERA	HON FA	CIORS									
	130	1.00	1.00	1.00	0.98	0.94	0.91	0.87	0.84	0.81	0.77	0.74	0.71	0.68
	120	1.00	1.00	1.00	1.00	0.96	0.92	0.89	0.85	0.82	0.79	0.76	0.73	0.70
AIR	110	1.00	1.00	1.00	1.00	0.98	0.94	0.90	0.87	0.83	0.80	0.77	0.74	0.71
то	100	1.00	1.00	1.00	1.00	0.99	0.96	0.92	0.88	0.85	0.82	0.78	0.75	0.72
TURBO	90	1.00	1.00	1.00	1.00	1.00	0.97	0.94	0.90	0.86	0.83	0.80	0.76	0.73
	80	1.00	1.00	1.00	1.00	1.00	0.99	0.95	0.92	0.88	0.85	0.81	0.78	0.75
(°F)	70	1.00	1.00	1.00	1.00	1.00	1.00	0.97	0.93	0.90	0.86	0.83	0.79	0.76
	60	1.00	1.00	1.00	1.00	1.00	1.00	0.99	0.95	0.91	0.88	0.84	0.81	0.78
	50	1.00	1.00	1.00	1.00	1.00	1.00	1.00	0.97	0.93	0.90	0.86	0.83	0.79
		0	1000	2000	3000	4000	5000	6000	7000	8000	9000	10000	11000	12000
						A	LTITUDE	FEET AB	OVE SEA	LEVEL)				
ΔFTF	RCOOL	ER HEA			ACTOR	S (ACH		1						
			TREOL		ACTON									
	130	1.51	1.59	1.67	1.76	1.85	1.93	1.93	1.93	1.93	1.93	1.93	1.93	1.93
	120	1.41	1.49	1.57	1.65	1.74	1.82	1.82	1.82	1.82	1.82	1.82	1.82	1.82
AIR	110	1.30	1.38	1.46	1.55	1.63	1.71	1.71	1.71	1.71	1.71	1.71	1.71	1.71
TO	100	1.20	1.28	1.36	1.44	1.52	1.60	1.60	1.60	1.60	1.60	1.60	1.60	1.60
TURBO	90	1.10	1.17	1.25	1.33	1.41	1.49	1.49	1.49	1.49	1.49	1.49	1.49	1.49
	80	1.00	1.07	1.15	1.22	1.30	1.38	1.38	1.38	1.38	1.38	1.38	1.38	1.38
(°F)	70	1.00	1.00	1.04	1.12	1.20	1.27	1.27	1.27	1.27	1.27	1.27	1.27	1.27
	60	1.00	1.00	1.00	1.01	1.09	1.16	1.16	1.16	1.16	1.16	1.16	1.16	1.16
	50	1.00	1.00	1.00	1.00	1.00	1.05	1.05	1.05	1.05	1.05	1.05	1.05	1.05
		0	1000	2000	3000	4000	5000	6000	7000	8000	9000	10000	11000	12000
						A	LTITUDE	(FEET AB	OVE SEA	LEVEL)				
						NOISE		1						
F		ELD MEC				NUISE								
100% Load Data				dB(A)			(dB)							
Free Field DISTANC Mechanical THE ENG			E EROM	3.2	-	4.0	73.8	79.3	82.8	88.8	89.8	87.3	84.3	81.3
		THE ENGINE (FEET)		22.9	-	4.0	68.5	72.5	76.5	77.5	80.5	77.5	73.5	68.9
				49.2	78	3.0	66.4	77.4	71.4	66.9	74.4	71.9	67.9	62.4
		DIOTANIO			10	9.4	106.5	108.5	105.8	107.8	102.5	102.8	97.8	87.5
		DISTANCE FROM THE ENGINE (FEET)		22.9	96	5.0	91.7	94.1	89.4	96.1	88.1	88.4	82.7	75.4
				49.2	89	9.4	85.1	87.5	82.8	89.5	81.5	81.8	76.1	68.8
					Overa	al SPL	63 Hz	125 Hz	250 Hz	500 Hz	1 kHz	2 kHz	4 kHz	8 khz

#### FUEL USAGE GUIDE:

This table shows the derate factor required for a given fuel. Note that deration occurs as the methane number decreases. Methane number is a scale to measure detonation characteristics of various fuels. The methane number of a fuel is determined by using the Caterpillar Methane Number Calculation program.

#### ALTITUDE DERATION FACTORS:

This table shows the deration required for various air inlet temperatures and altitudes. Use this information along with the fuel usage guide chart to help determine actual engine power for your site.

#### ACTUAL ENGINE RATING:

It is important to note that the Altitude/Temperature deration and the Fuel Usage Guide deration are not cumulative. They are not to be added together. The same is true for the Low Energy Fuel deration (reference the Caterpillar Methane Number Program) and the Fuel Usage Guide deration. However, the Altitude/Temperature deration and Low Energy Fuel deration are cumulative; and they must be added together in the method shown below. To determine the actual power available, take the lowest rating between 1) and 2).

- 1) (Altitude/Temperature Deration) + (Low Energy Fuel Deration)
- 2) Fuel Usage Guide Deration

Note: For NA's always add the Low Energy Fuel deration to the Altitude/Temperature deration. For TA engines only add the Low Energy Fuel deration to the Altitude/Temperature deration is less than 1.0 (100%). This will give the actual rating for the engine at the conditions specified.

#### AFTERCOOLER HEAT REJECTION FACTORS (ACHRF):

Aftercooler heat rejection is given for standard conditions of 77°F and 500 ft altitude. To maintain a constant air inlet manifold temperature, as the air to turbo temperature goes up, so must the heat rejection. As altitude increases, the turbocharger must work harder to overcome the lower atmospheric pressure. This increases the amount of heat that must be removed from the inlet air by the aftercooler. Use the aftercooler heat rejection factor (ACHRF) to adjust for ambient and altitude conditions. Multiply this factor by the standard aftercooler heat rejection. Failure to properly account for these factors could result in detonation and cause the engine to shutdown or fail.

#### SOUND DATA:

Data determined by methods similar to ISO Standard DIS-8528-10. Accuracy Grade 3. SPL = Sound Pressure Level.

# G3406

GAS ENGINE TECHNICAL DATA



## NOTES

- 1 ENGINE RATING IS WITH 2 ENGINE DRIVEN WATER PUMPS. TOLERANCE IS ± 3% OF FULL LOAD.
- 2 GENERATOR POWER DETERMINED WITH AN ASSUMED GENERATOR EFFICIENCY OF 91.9% AND POWER FACTOR OF 0.8 [GENERATOR POWER = ENGINE POWER x GENERATOR EFFICIENCY].
- 3 ISO 3046/1 ENGINE EFFICIENCY TOLERANCE IS (+)0, (-)5% OF FULL LOAD % EFFICIENCY VALUE. NOMINAL ENGINE EFFICIENCY TOLERANCE IS  $\pm$  5% OF FULL LOAD % EFFICIENCY VALUE.
- 4 THERMAL EFFICIENCY: JACKET HEAT + LUBE OIL HEAT + EXH. HEAT TO 350°F.
- 5 TOTAL EFFICIENCY = ENGINE EFF. + THERMAL EFF. TOLERANCE IS ± 10% OF FULL LOAD DATA.
- **6** ISO 3046/1 FUEL CONSUMPTION TOLERANCE IS (+)5, (-)0% OF FULL LOAD DATA. NOMINAL FUEL CONSUMPTION TOLERANCE IS  $\pm$  5 % OF FULL LOAD DATA.
- 7 UNDRIED AIR. FLOW TOLERANCE IS  $\pm$  5 %
- 8 INLET MANIFOLD PRESSURE TOLERANCE IS ± 5 %
- **9** INLET MANIFOLD TEMPERATURE TOLERANCE IS ± 9°F.
- **10** TIMING INDICATED IS FOR USE WITH THE MINIMUM FUEL METHANE NUMBER SPECIFIED. CONSULT THE APPROPRIATE FUEL USAGE GUIDE FOR TIMING AT OTHER METHANE NUMBERS.
- 11 EXHAUST STACK TEMPERATURE TOLERANCE IS (+)63°F, (-)54°F.
- **12** WET EXHAUST. FLOW TOLERANCE IS ± 6 %
- 13 NOX VALUES ARE "NOT TO EXCEED".
- 14 CO, CO2, THC, and NMHC VALUES ARE "NOT TO EXCEED".
- 15 O2% TOLERANCE IS  $\pm 0.2$ .
- **16** LHV INPUT TOLERANCE IS ± 5%.
- 17 HEAT REJECTION TO JACKET TOLERANCE IS ± 10 % OF FULL LOAD DATA, BASED ON TREATED WATER.
- 18 HEAT REJECTION TO ATMOSPHERE TOLERANCE IS ± 50% OF FULL LOAD DATA, BASED ON TREATED WATER.
- **19** HEAT REJECTION OF LUBE OIL TOLERANCE IS ± 20% OF FULL LOAD DATA, BASED ON TREATED WATER.
- 20 HEAT REJECTION TO EXHAUST TOLERANCE IS ± 10% OF FULL LOAD DATA, BASED ON TREATED WATER.
- 21 HEAT REJECTION TO A/C TOLERANCE IS  $\pm\,5$  % OF FULL LOAD DATA, BASED ON TREATED WATER.

### SITE SPECIFIC COOLING SYSTEM SIZING EQUATIONS (WITH TOLERANCES)

- **22** TOTAL JACKET CIRCUIT  $(JW+OC) = (JW \times 1.1) + (OC \times 1.2)$ .
- **23** TOTAL AFTERCOOLER CIRCUIT (AC) = AC x ACHRF x 1.05.

DM0804-01

PAGE 3 OF 3

21-Sep-09